

Analyzing the Numerical Behavior of a Vibration System with Non-Linearity Using Single Degree of Freedom Approach

M. Ebrahim

Automotive and Marine Dept., College of Technological Studies, PAAET, Kuwait

Abstract A numerical analysis of a Single Degree of Freedom Vibration (SDOF) vibration system with non-linearity is investigated. Mag-spring as a nonlinear spring with hardening behaviour is added in parallel to a vibrating system. The nonlinear effect is expected to curve the vibration amplitude and shift away the system's natural frequency only where it is necessary. Due to hardening behaviour of the nonlinear Mag-spring especially at higher amplitudes, nonlinearity kicks in and the stiffness of the system is increased; then the natural frequency of the system is shifted to a higher value away from the excitation frequency. It has been shown that how a light damping could lead a forced vibration system to a steady state. Moreover, it explains how a nonlinear spring can be fitted in parallel to the main system with one degree without changing its degree of freedom. The techniques utilised to formulate the equation of non-linear spring have been explained as well. Based on the characteristics of the nonlinear spring, a system with appropriate parameters has been formulated. Further, a second order nonlinear differential equation is solved by a written program in MATLAB software. It shows how the initial conditions and the differential equation in a state space form will be used by ode45 function in MATLAB to solve the system.

Keywords Single degree of freedom, Vibration, Mag-spring, MATLAB

1. Introduction

High line speeds can result in thermal distortion in the shift clutch of the vehicle transmission system, which will generate vibration and have a significant impact on the performance of the friction element and other related components. The design of the clutch of a vehicle power-shift steering transmission system depends heavily on the thermal deformation of the friction pairs and its vibration. As a result, it has important theoretical and engineering ramifications to thoroughly examine the vibration appearances of the friction pairs at high line speed and to advance the development of a practical vibration control system.

A microscopic normal element contact model of thermal distortion is recognised by appertaining the theory and technology of surface structure characteristics and contact mechanics of friction pairs from the belvedere of micro mechanics. The link between micro and macro is then established, making full use of statistical statistics and standardisation techniques. Thus, a mathematical model of macro contact between two pairs of frictional pairs is

obtained. The viscoelastic contact property including stress and strain is constructed at the same time as the Kelvin-Voigt model is presented and the viscoelastic contact differential operator is developed into a mathematical model, allowing for the creation of a viscoelastic contact mathematical model of the friction pairs. Finally, mechanical analysis yields the mathematical model of nonlinear vibration, and its vibrational features are simulated. The accuracy of the simulation model is confirmed by testing the nonlinear vibration characteristics with varying rotational speed and lubricant. By adopting a magnetorheological (MR) fluid damper, key, and subharmonic concurrent resonance for the Duffing-type nonlinear system under base innervation and external innervation is passively controlled. The fractional-order derivative Bingham model of MR fluid damper is assessed in this study. The incremental averaging method is used to arrive at the system's approximated logical solution.

Based on gaining the principal resonance of the system under base innervation by the averaging method, the subharmonic resonance solution of the system is obtained by taking the subharmonic resonance of the system under base innervation and external innervation as an increase, to obtain the estimated analytical solution of the concurrent resonance of primary and subharmonic resonance. And the amplitude–frequency equation and phase–frequency equation of the steady-state solutions for the primary and

* Corresponding author:

m.ebrahim@q8scholar.com (M. Ebrahim)

Received: Apr. 20, 2023; Accepted: May 10, 2023; Published: Jun. 17, 2023

Published online at <http://journal.sapub.org/jmea>

subharmonic resonance of the system are derived respectively. According to the approximate analytical solutions, the steadiness conditions of the steady-state solution of the primary resonance and subharmonic instantaneous resonance are obtained by Lyapunov method. Hence, it becomes necessary to study latest improvements in oscillating isolators applied in various applications.

A review has been carried out by scholars Alkhatib & Golnaraghi [1] to achieve structural vibration control. Several scholars—namely Symans & Constantinou [2], Chu et al. [3] and Casciati et al. [4] have performed research efforts concerning the development and utilisation in semi-active structural control system for vibration reduction applications. Passive control systems have undergone various changes in order to enable adjustments to be made in regard to the mechanical aspects; these are considered to be the benchmarks from which semi-active systems originate. Semi-active vibration control systems offer most performance characteristics of active systems, and they don't require large power sources. Moreover, they are not a threat to destabilize the system, and that because they don't inject energy into the controlled system [2,5,6].

A comprehensive coverage of the theory of linear vibration isolation was provided by Crede [7], Snowdon [8] and Rivin [9]. Tuned Vibration Absorbers (TVA) was pioneered by Watte [10] and Frahm [11]. As stated by researchers, by increasing the damping ratio, working range of the main system becomes wider (positive point), however the magnification ratio increases negative point as well. Since, the first mathematical model was proposed by Ormondroyd and Den Hartog [12], huge investigations have been carried out in the past years to introduce new models for passive TVAs that could solve the off-tuning problems. Among the latest, Liu and Coppola [13], offered an optimal design for a damped linear vibration absorber including a new combination of the mass and damping as well. The fundamental idea behind vertical and horizontal motions were estimated by Winterflood et al. [14]. Dynamic anti-resonant vibration isolators are another class of linear isolator introduced in the literature by Goodwin [15], Halwes [16] and Flannelly [17]. Goodwin and Halwes [15,16] emphasise that anti-resonant isolators are based on hydraulic leverage, although the isolator of Flannelly [17] is based on a levered mass. The levered mass lowers the resonant frequency of the isolator, which accordingly enhances its overall ability to operate at a lower frequency range. This type of anti-resonant is used in an isolation helicopter rotor and floor machines [18-21]. The rubber engine mounts were replaced with hydraulic engine mounts by Corcoran & Ticks [22] and Flower [23] and comprise what is referred to as inertia track property; notably, there are several similar components between the frameworks of Goodwin and Halwes [15,16]. Yilmaz & Kikuchi [24,25] investigated the theory of anti-resonant vibration isolators, comprising various degrees of freedom regarding passive isolators comprising at least two anti-resonance frequencies.

The effect of a non-linear isolator on the communicable,

relays in particular regard on its inflexibility if it is flexible or firm [26]. when considering the suspensions' response regarding high-speed vehicle and fragile instruments' mounts—non-linearity becomes essential to be considered [27]. Hundal & Parnes [28] studied a vibration system when under base oscillation. Moreover, Metwalli [29] introduced a design centred on enhancing and improving the suspension systems non-linear in nature, that accordingly established to perform better than their opposite linear systems. Nayfeh et al. [30] and Yu et al. [31] investigated mechanical passive vibration isolators, incorporating distinct damping, mass, and stiffness aspects, highlighting that the more confined non-linear styles may be incorporated within the system through the proper design of the system's stiffness non-linearities. Popov & Sankar [32] proposed a non-linear orifice type damping comprising a changing damping coefficient, which is regarded as an interior geometry function, flow fluctuation frequency, and Reynolds number.

Vibration control methods have been classified into three areas, namely passive, active or hybrid (active/passive) methods. Overall, the techniques accepted for vibration control in the context of industrial equipment comprise damping, force reduction, mass addition, isolation, and tuning the system's natural frequency. Many scenarios and case studies have been introduced, with emphasis placed on the tuning of the natural frequency as a method to avoid resonance or postpone from the working frequency of the system.

Latest developments in metallic and viscoelastic non-linear vibration isolators have been provided with covering classical and non-classical systems. The major characteristics of non-linear isolators were explained. These characteristics contain the shifting of the resonance frequency, jump phenomena, chaotic motion, and internal resonance. The transmissibility curves of non-linear isolators subjected to non-harmonic force require explanation regarding the mean squares of response, as well as excitation amplitudes. Viscoelastic non-linear isolators offer various mechanical features owing to their stiffness and damping parameters dependence on frequency and temperature. As has been well documented, the vibration amplitude of the non-linear viscoelastic isolator is known to increase in line with the reduction of the response amplitude, and the transmissibility is improved over that of the linear isolator for vibration frequency, which exceeds a specific value governed by the temperature and excitation amplitude.

The literature survey has shown that non-linearity has been utilised widely in terms of controlling vibration by many people. Therefore, this research will investigate non-linearity in vibration control. Several examples have been brought from the literature. The major element noticed in using non-linearity in the past was the fact that, these non-linearities were mechanically designed, structurally fixed, and difficult to modify. Therefore, designing a non-linear spring to give vibration characteristics was a very difficult task.

However, the developments in regard of the magnetic

spring technology have supported the idea that non-linearity can now be designed to specifications using magnetic technology. Therefore, the idea of studying the use of magnetic technology in terms of developing a non-linear spring, and using it in vibration control, is a thought that has come to mind.

This research proposes a new idea that has the potential to lead to overcoming the previously mentioned shortages. The improvements in terms of magnetic technology have encouraged this research with the new invented Mag-spring (magnetic spring explained in the next chapter). The Mag-spring was invented for a purpose other than vibration cancellation. However, the non-linearity that Mag-spring shows brought the idea of investigating a new way that could guide to a novel point of view in enhancing the vibration cancellation in vibrating structures.

2. Research Methodology

Hardening non-linearity is used for design of vibration based on single degree of freedom system. the effect of non-linearity on vibration characteristic was studied here. Mag-spring was chosen to represent the non-linearity in the system. Mag-spring is a totally passive device that consists of two permanent magnets (introducing the attractive force) as a slider and a stator sliding against each other while separated and guided by a plain Teflon as a bearing as shown in Figure 1. To investigate the effect of nonlinearity on vibration characteristic the following steps have been considered:

1. Prepare a mathematical model to simulate nonlinearity of the new invented mag-spring.
2. validate the numerical simulation to find out what kind of expectations are there.

3. Results and Discussion

3.1. Load Deflection Experiment of the Spring

The rational of selecting a nonlinear spring is explained above and as it will be demonstrated, the spring will provide

the required behaviours both for hardening and softening. To start developing the simulation model, the stiffness curve of the spring will be used in mathematical models from the outset. In the experiment, the applied loads are varied to observe the extension of the spring. For every applied load, the extension is recorded and tabulated as shown in the Table 1. Based on recorded data, a non-linear regression equation has been established using curve-fitting technique.

Table 1. Amplitude vs load in calibrating the non-linear magnetic spring

Amplitude (X[m])	Load [N]
0	0
0.00126	1
0.00217	2
0.00323	2.5
0.00445	3
0.00562	5.4
0.00712	8.2
0.00853	11
0.00958	17
0.01027	21
0.0118	25
0.0128	28.5
0.0138	33
0.0151	36
0.0164	38
0.0178	40
0.019	42
0.02015	42
0.02118	43
0.0222	43.9
0.02337	43.5
0.0247	44
0.0258	44

The result of the MATLAB computational function produces a non-linear model of the magnetic spring as follows:

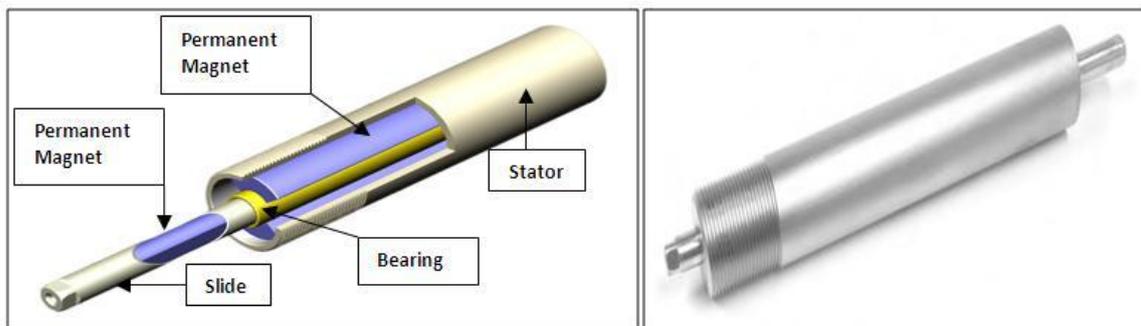


Figure 1. Magnetic spring to provide non-linearity (www.linmot.com)

$$\begin{aligned}
 F_m = & 5.62E + 18X^{11} + 5.27E + 16X^{10} - 7.68E + 16X^9 \\
 & -9.91056E + 13X^8 + 1.32305E + 14X^7 \\
 & +64735435094X^6 - 87047721025X^5 \\
 & -16445112.06X^4 + 22982840.36X^3 \\
 & +1353.713999X^2 + 288.4233002X \\
 & -0.015560957
 \end{aligned}
 \quad (1)$$

This equation is then fitted back to the graph and compared with the measured data and both curves are plotted in Figure 2. It is obvious that the trend of the elongation of the magnetic spring is non-linear for the whole range of the loads applied. However, after 40N the spring shows a kind of constant behaviour.

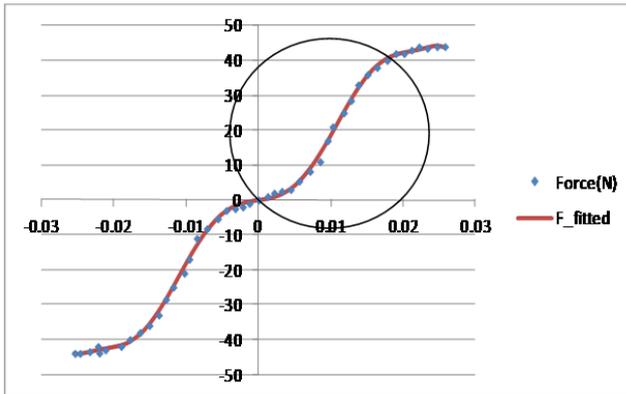


Figure 2. Fitting of Curve

Based on the characteristic of the nonlinear spring, a proper numerical model has been designed in Matlab software. In this study the effects of the nonlinear hardening spring have been conducted on a system with one degree of freedom. In this program the differential equation of the vibration with nonlinear stiffness has been solved numerically and the results of the analysis has been shown in time and frequency domain.

The beam in the investigated vibration system (Figure 3) is considered as a leaf spring to represents the linear spring in constructing the single degree of freedom system. This is chosen because in practical terms it gives an easier system to construct. Then the stiffness of the leaf spring is calculated based on the simple beam theory. The system's response will be compared with linear spring and then with nonlinear spring in parallel with the linear spring.

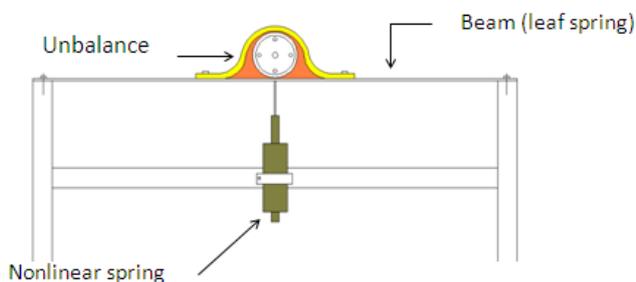


Figure 3. Proposed vibration system

3.2. Numerical Analysis of Non-Linear System

Numerical analysis is divided into mainly five subsections as follows:

1. The behaviour of one degree of freedom forced vibration system. It has been shown that how a light damping could lead a forced vibration system to a steady state. Moreover, it will explain how a nonlinear spring will be fitted in parallel to the main system with one degree without changing its degree of freedom. The techniques utilised to formulate the equation of non-linear spring have been explained as well. Based on the characteristics of the nonlinear spring, a system with appropriate parameters has been formulated.
2. Program in MATLAB software for second order nonlinear differential equation.
3. A parametric study to choose a proper main system with specific forced vibration has been discussed in section.
4. Simulation results for a harmonic excitation with constant amplitude have been explained in this section.
5. The behaviour of the system under a forced excitation due to the unbalance mass.

3.3. Forced Vibration of a Linear System with One Degree of Freedom

The behaviour of a forced vibration system with one degree of freedom is to be discussed and its parameters will be explained. At the beginning, the analytical solution of a linear system has been studied as a reference case. Also, the necessity of light damping in a forced vibration system to achieve steady state solution has been discussed. In addition, forced vibration of a nonlinear system has been formulated as well. Moreover, it has been shown that, how curve fitting is used to formulate the nonlinear behaviour (stiffness) of the system with a polynomial of the order 11. Equation 2 is the differential equation for a linear system with SDOF. The solution to this equation contains two parts, the complementary (homogenous) function and the solution due to the forced function, $F(t)$.

$$M\ddot{x} + c\dot{x} + kx = F(t) \text{ where } F(t) = F_0 \cos\omega t \quad (2)$$

$$\begin{aligned}
 x(t) = & e^{-(c/2m)t} \left(A e^{\frac{(c/2m)^2 - k/m}{m}t} + B e^{-\frac{(c/2m)^2 - k/m}{m}t} \right) \\
 & + \frac{F_0 k}{1 - (\omega/\omega_n)^2 - \frac{c^2}{4k}} \cos(\omega t - \phi) \quad c \neq 0
 \end{aligned}
 \quad (3)$$

$$\begin{aligned}
 x(t) = & A \sin\omega_n t + B \cos\omega_n t + \frac{F_0/k}{1 - (\omega/\omega_n)^2} \cos\omega t \quad \text{IF } c \\
 = & 0, \text{ without damping}
 \end{aligned}
 \quad (4)$$

$$\text{where } \omega_n = \sqrt{\frac{k}{M}} \text{ and } r = \frac{\omega}{\omega_n} \quad \phi = \tan^{-1} \frac{c\omega}{k - m\omega^2}$$

The unknown coefficient in the complementary part of equations 3 and 4, A and B , are determined by forcing the

initial conditions. Applying initial condition for the undamped system:

Finally,

At $t = 0, x = \dot{x} = 0$ then $A = 0$ and $B = -\frac{F_0/k}{1-r^2}$

$$x(t) = \frac{F_0/k}{1-r^2} (\cos \omega t - \cos \omega_n t) \text{ or} \tag{5}$$

$$x(t) = \frac{F_0/k}{1-r^2} \sin\left(\frac{\omega_n + \omega}{2}\right) t \sin\left(\frac{\omega_n - \omega}{2}\right) t$$

Equation 5 shows the response of an undamped linear system with forced vibration. As it can be seen from the equation, this response has two harmonic oscillations with different frequencies. Vibrations with periodically changing amplitude resulting from the superposition of two harmonic oscillations (with close frequencies) is called beat. Beats arise because the phase difference between two vibrations with different frequencies (but near) is constantly varying such that both vibrations are in phase at one moment, after some periods of time are out of phase, and then again in phase, and so on. In other word, the component with lower frequency will cover the higher frequency components likes an envelope and this can be seen as beating phenomena when two frequencies are close to each other (Figure 4, left). Depend on the frequency of excitation, two different conditions could happen. In the case of $\omega_n \neq \omega$ or $r \neq 0$, the beating phenomena (Figure 4, left) with the beating period of $2\pi/(\omega_n - \omega)$ will happen. However, when $\omega_n = \omega$ or $r = 1$, the beating frequency and amplitude go to infinity (Figure 4, right). The vibration amplitude keeps increasing and this phenomenon is called resonance. This is the reason when designing a system, the system's natural frequency should be estimated with keeping in mind that excitation frequency should not go near the natural frequency of the system. Therefore, shifting the natural frequency away from the excitation frequency is done by adding nonlinear stiffness to the main system.

In the previous section, the resonance phenomena has

been shown for an ideal undamped system. However, this is a very idealistic situation, because in nature, it is impossible to make a system without energy dissipation. Therefore, to make the analysis more realistic, a very light damping is introduced while the system remains notionally undamped. The response of the forced vibration damped system (Eqn. 4) has been analysed to define a light damping coefficient. The term $e^{-(c/2m)t}$ represents the exponential decaying function of time. However, the behaviour of the term $(Ae^{\sqrt{(c/2m)^2 - k/m}t} + Be^{-\sqrt{(c/2m)^2 - k/m}t})$ depends on the numerical value of the $((c/2m)^2 - k/m)$ (under the radical); if it is positive, negative or zero. When damping term is larger than k/m , the radical become positive, then exponents in the Eqn. 4 become real and no oscillation is possible. This situation is called overdamped response. On the other hand, when damping term $(c/2m)^2$ is smaller than k/m , the radical become negative, then exponents in the Eqn. 4 become imaginary number and oscillations are possible. This situation is called underdamped response. If damping term $(c/2m)^2$ is equal to the k/m , radical become zero. This situation is called critical damping, c_c response:

$$c_c = 2m \sqrt{\frac{k}{m}} = 2m\omega_n = 2\sqrt{km} \tag{6}$$

Any damping can then be expressed in terms of the critical damping by the non-dimensional number ζ , which is called damping ratio, $\zeta = \frac{c}{c_c}$.

3.4. Light Damping

According to Figure 5, for the cases where the damping ratio is less than $\zeta < 0.1$ although the system is underdamped, the damping does not have much effect on the response of the system. For instance, if the system with mass, $m = 1$, Stiffness $k = 10000\text{N.m}$, natural frequency is $\omega_n = 100\text{rad/s}$; if we define damping ratio $\zeta = 0.01$ and $cc = 2m\omega_n = 200$; then damping coefficient, c , in the Eqn.4.1 becomes $c = 2$.

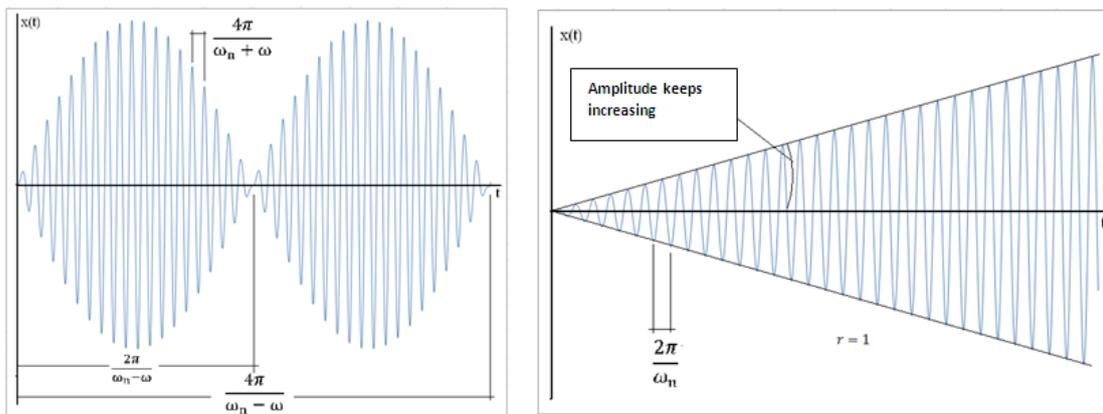


Figure 4. Response of a linear single degree of freedom system with forced vibration. Left, frequency of excitation is different from natural frequency of the system ($\omega_n \neq \omega$). Right, frequency of excitation and natural frequency of the system are equal ($\omega_n = \omega$)

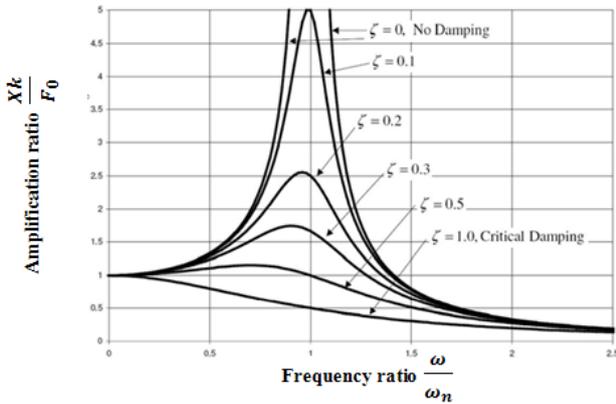


Figure 5. Diagram of Amplitude ratio versus Frequency ratio for a forced vibration system - the response of system at different damping ratio (Tustin 2006)

Figure 6 shows how defining a light damping ($\zeta=0.01$) could lead the system to a steady state situation after passing a transient part. A comparison between Figure 6-left and Figure 6-right shows that by introducing a light damping, the system vibrates with the frequency of excitation force. In this case the response of the system will change to the following equation:

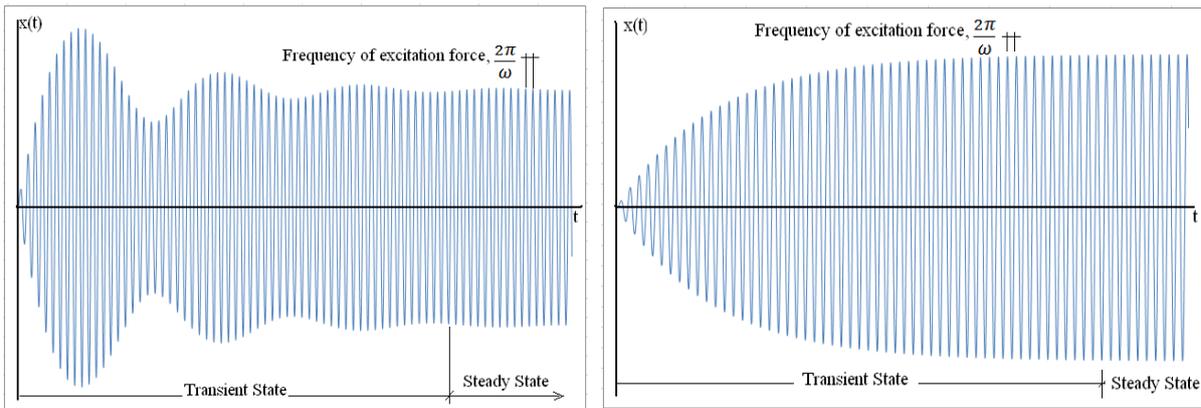


Figure 6. Light damping on a system with forced vibration, left ($\omega_n \neq \omega$), Right ($\omega_n = \omega$)

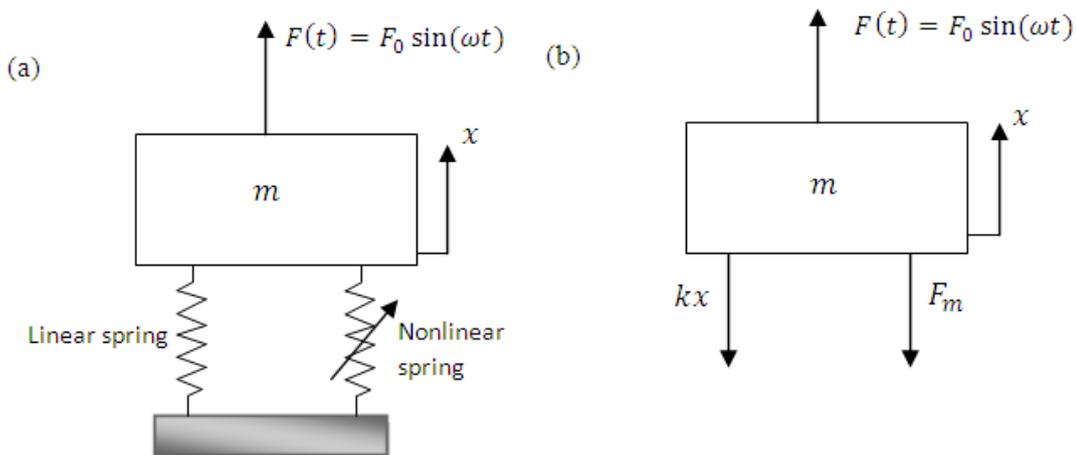


Figure 7. (a) nonlinear spring was added in parallel to a system with one degree of freedom. (b) Free body diagram of the system with nonlinear stiffness

$$x(t) = \frac{F_0/k}{1-r^2} \cos\omega t \tag{7}$$

3.5. Forced Vibration of a Nonlinear System with One Degree of Freedom

The schematic diagram of the system with nonlinear spring is shown in Figure 7. Nonlinear spring was added in parallel to the system. The analysis is a forced vibration type with the frequency of excitation of ω .

Free body diagram of the system has been shown in Figure 7(b). The force conducted by nonlinear spring is shown by F_m ; The force created by the linear spring is kx . Based on Figure 4-4 (b), the equation of the system can be derived as follows:

$$\begin{aligned} \sum F &= m\ddot{x} \\ F(t) - kx - F_m &= m\ddot{x} \\ m\ddot{x} + kx + F_m &= F(t) \end{aligned} \tag{8}$$

Equation 8 is a nonlinear differential equation due to the force, F_m . To solve the equation a numerical method has been used. The use of MATLAB to compute the solution of the equation and the results of the simulation will be shown in the following section.

3.5.1. Behavior of SDOF Vibration System

Single degree-of-freedom (SDOF) systems are widely used in engineering applications to model the dynamic response of structures to external forces or excitations. These systems are characterized by a single mass, a single degree of freedom, and a linear restoring force that represents the stiffness of the system. However, many real-world structures exhibit non-linear behavior due to factors such as material nonlinearity, geometric nonlinearity, or damping nonlinearity.

The behavior of SDOF vibration systems with non-linearity is of great practical importance in engineering and science. In fact, the understanding of the behavior of such systems is crucial for designing structures that can withstand external excitations such as earthquakes, wind, and other dynamic loads.

Some practical applications of non-linear SDOF vibration systems include:

1. Seismic design of buildings: Earthquakes are one of the most destructive forces that a building can experience. SDOF models are commonly used to analyze the response of structures to seismic loads. However, non-linearities such as yielding of materials or plastic deformations can significantly affect the behavior of the structure and, therefore, the design process. Understanding the behavior of SDOF systems with non-linearity is, therefore, critical for ensuring the safety and reliability of buildings under seismic loads.
2. Aerospace engineering: Non-linear vibrations can be encountered in various aerospace applications, such as aircraft and satellite structures. Understanding the behavior of these systems is crucial for designing safe

and reliable structures that can withstand the extreme environmental conditions in space and during launch.

3. Mechanical engineering: Non-linearities can also arise in mechanical systems such as engines, turbines, and gearboxes. The understanding of the behavior of SDOF systems with non-linearity is essential for predicting the performance and durability of these systems and, therefore, for improving their design and efficiency.
4. Biomedical engineering: The human body is a complex system with many non-linearities, and the understanding of the behavior of SDOF systems with non-linearity is crucial for designing medical devices such as prosthetics, implants, and diagnostic tools.

The behavior of SDOF vibration systems with non-linearity has significant practical applications in a wide range of engineering and scientific fields. The understanding of the behavior of such systems is essential for designing safe and reliable structures and devices, improving their efficiency, and ensuring the safety and well-being of people and their environment.

3.5.2. Curve Fitting

The curve fitting equation that will be tested in the model of the non-linear vibration system is the full polynomial equation obtained by curve fitting the data points using matlab function polyfit.

This nonlinear spring force equation is represented as F_m in non-linear vibration system model. The nonlinear differential equation does not have an analytical solution, hence, to find the response of this equation in time and frequency domain a program in MATLAB software was prepared.

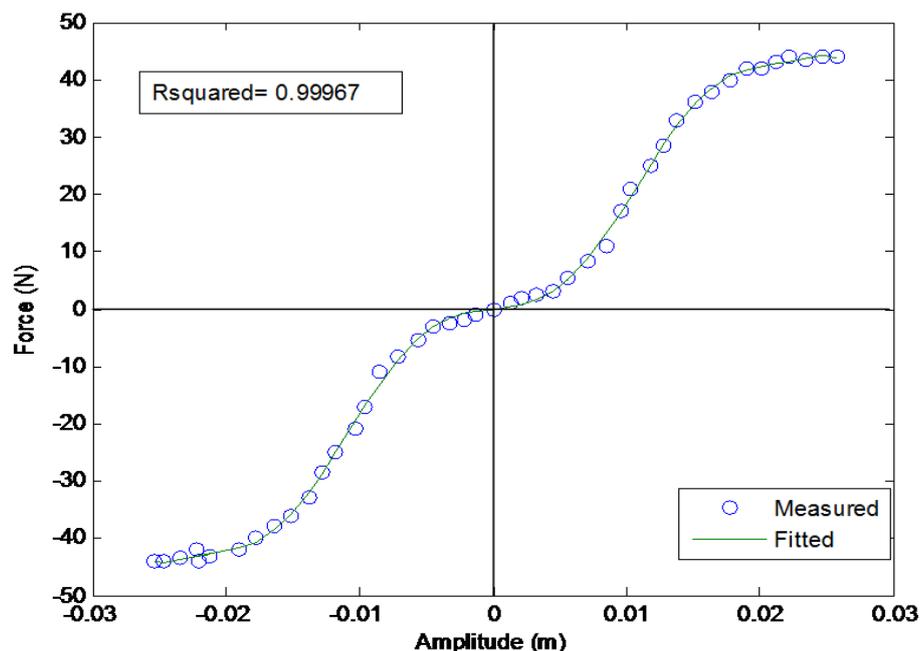


Figure 8. Curve fitting

3.5.3. MATLAB Software

The equation 8 has been developed to model a SDOF vibration system. This is a nonlinear second order differential equation and there is no straight forward analytical solution for this equation. Therefore, in this work, the solution for this equation will be computed in a numerical analysis with the aid ODE45 function in MATLAB Software.

3.5.4. ODE45 Function

To solve the non-linear vibration equation obtained and displayed in equation 8, a separate function of $F(t,x)$ is built and this function will be called an ordinary differential equation solver, ODE45 using MATLAB. This routine utilizes a variable step fourth order Runge-Kutta Method with 5th order corrector in solving differential equations numerically as follows: $[t,x]=ode45(@my_function,[t0,tf],[x0,xdot0])$, where "my_function" is a Matlab function $F(t,x)$, which has been used to introduce the nonlinear differential equation while, $t0$ introduces the initial time, the final time is tf and $x0$ is the initial displacement and the initial velocity is $xdot0$. The results of the differential equation are stored in an array called x , which contains in its first column the values of amplitudes and in the second column the corresponding values of velocities, $x[x, xdot]$.

3.5.5. Initial Condition

The system is assumed to start from the rest position, where displacement $x0 = 0$ and initial velocity, $x = 0$. The initial condition should not be out of the operating range.

```
%initial conditions
x0=0; %displacement
xdot_0=0; %velocity
```

3.5.6. State Space Variables

To use the ode45 function, it is essential to write the differential equation in state space form.

```
xdot1=x(2);
xdot2=-k1/m1*x(1)-1/m1*(a11*x(1)^11+a10*(x(1))^10+
a9*(x(1))^9+a8*(x(1))^8+a7*(x(1))^7+...
a6*(x(1))^6+a5*(x(1))^5+a4*(x(1))^4+a3*(x(1))^3+a2*(x
(1))^2+a1*(x(1))+a0)+1/m1*F0*sin(w*t)-ldamp*x(2)/m1;
xdot=[xdot1;xdot2];
```

In the above functions the $x(1)$ and $x(2)$ are the x array components, which shown displacement and velocity, respectively. $Xdot1$ and $xdot2$ are the first and the second differentiation of the displacement (velocity and acceleration).

3.5.7. Frequency Domain

Among the key code in programming the model of the vibration system is to determine the maximum amplitude for every frequency set. Since the time constants are dependent to the iterations done in ODE45 function, the array of the highest amplitude varies from one frequency to another.

Therefore, a dynamic loop is developed to search the highest value in an array and stores it in an exclusive maximum amplitudes array, $x_n(i)$, and this array can be used to plot a graph for maximum amplitude against frequency later. The time of when this maximum amplitude happens is also recorded for verification. The important issue is to make sure the maximum amplitudes are selected from the steady state parts of the time domain. To find the steady state response it is possible to check the convergence of the results to meet a minimum difference between two consecutive maximum. Also, it is possible to let the program runs for a longer time; then to pick the maximum after a specific time which certainly the results have been converged. In this program, it has been checked that the steady state will happening after 5s in real time domain which is equivalent of 2500 steps in time (in ode45 function $dt=0.002s$). The code to search the highest amplitude in an array is written below.

```
%Determine the amplitude of every  $\omega$ 
```

```
max=0;
for b=2500:length(y)
x_b=(x{i}(b));
if x_b>max
max=x_b;
j=time(i);
end
end
x_n(i)=max;
```

3.5.8. Parametric Study

To show how nonlinearity could improve the behaviour of a system with one degree of freedom at a forced vibration system (especially when the excitation frequency is near the natural frequency of the main system), a spring with nonlinear behaviour has been introduced. This nonlinear spring has a hardening/softening behaviour. At low amplitude of vibration this spring shows hardening behaviour, which means at the beginning it has very low stiffness and gradually it will increase by increasing the amplitude of vibration. As a result, the nonlinear spring at low amplitude of vibration does not affect the main system very much which is desirable. On the other hand, when the amplitude of vibration is increased (resonance) the nonlinear spring shows higher stiffness and shifts the natural frequency of the system out of the excitation frequency range.

3.5.9. Defining the Main System Parameters

To choose a main system (stiffness and mass) which is suitable for our nonlinear spring, a series of runs have been performed. At the first stage, a system with natural frequency $\omega_n = 100rad/s$ was chosen and the nonlinear spring was added to the system (in parallel); then a set of runs were defined. The stiffness and mass of the main system were changing in a way to keep constant natural frequency of the

main system. Figure 9 shows that the nonlinear stiffness is more effective for the system with lower stiffness and mass at constant natural frequency $\omega_n = 100 \text{ rad/s}$. This could be predicted as the nonlinear spring will add to the stiffness of the main system and the system becomes stiffer (without

increasing the mass of the system); this issue is not recommended for the frequencies lower than natural frequency of the main system as to make the system stiffer were it is not really necessary, however it will help to shift the resonance to higher frequencies which is desirable.

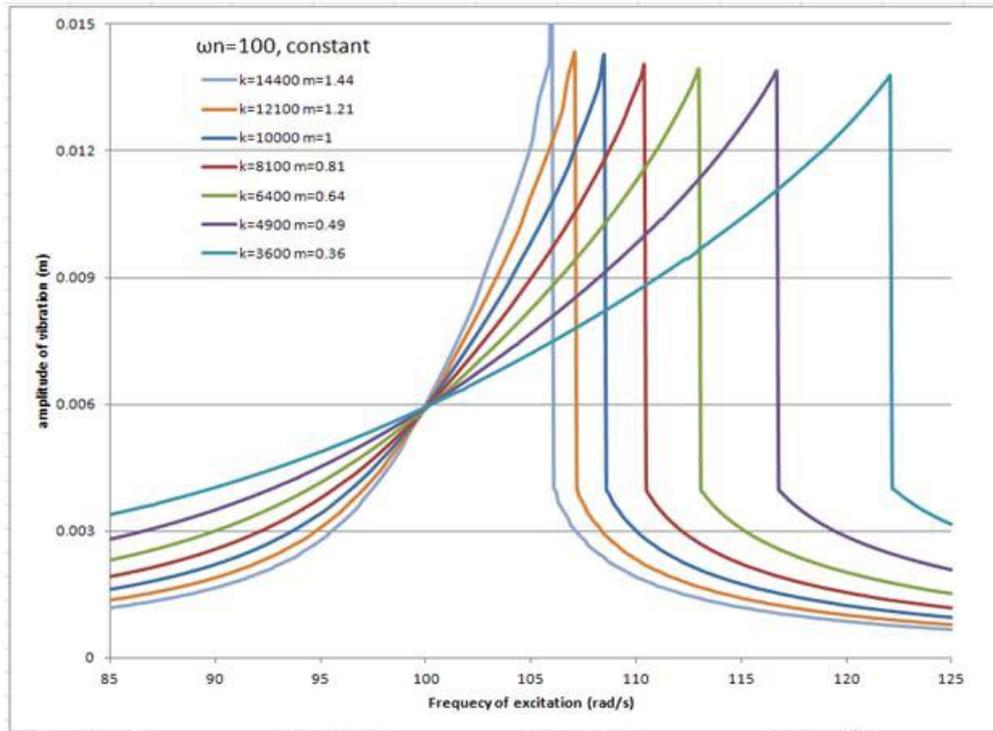


Figure 9. Amplitude of vibration verses frequency of excitation (frequency domain) for constant natural frequency of the main system with different stiffness and mass

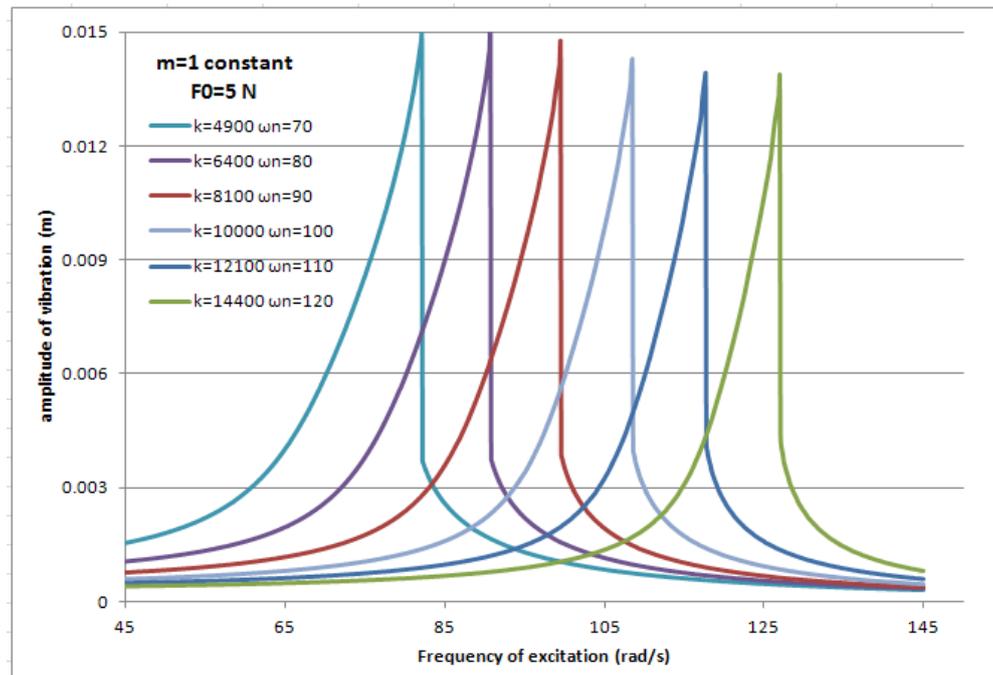


Figure 10. Amplitude of vibration verses the frequency of excitation. Mass of the main system is constant however the stiffness of the main system is different for each set of data

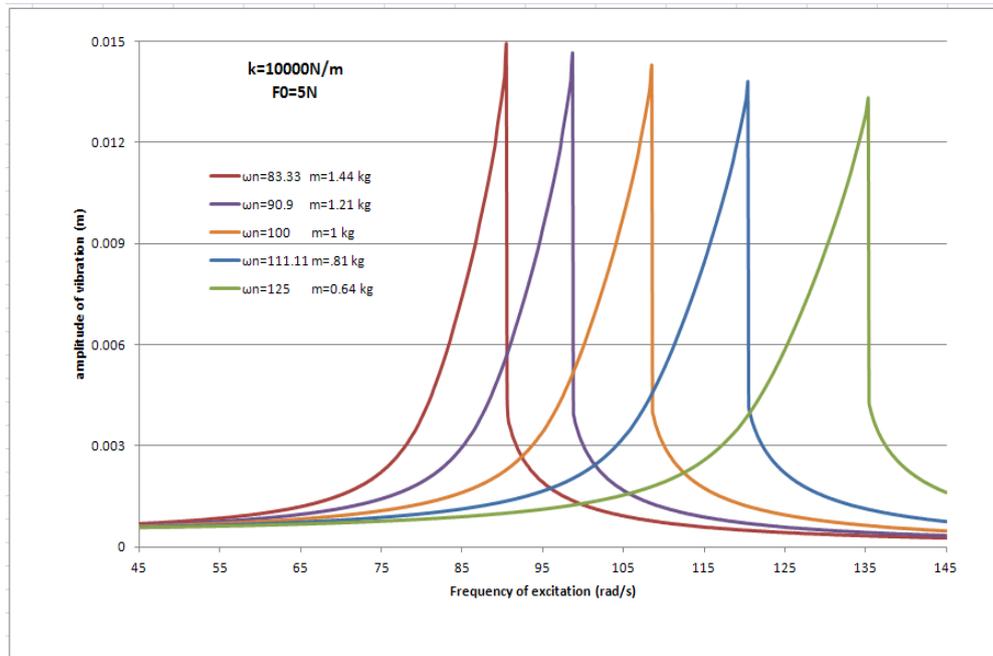


Figure 11. Amplitude of vibration versus the frequency of excitation. Stiffness of the main system is constant however the mass of the main system is different for each set of data

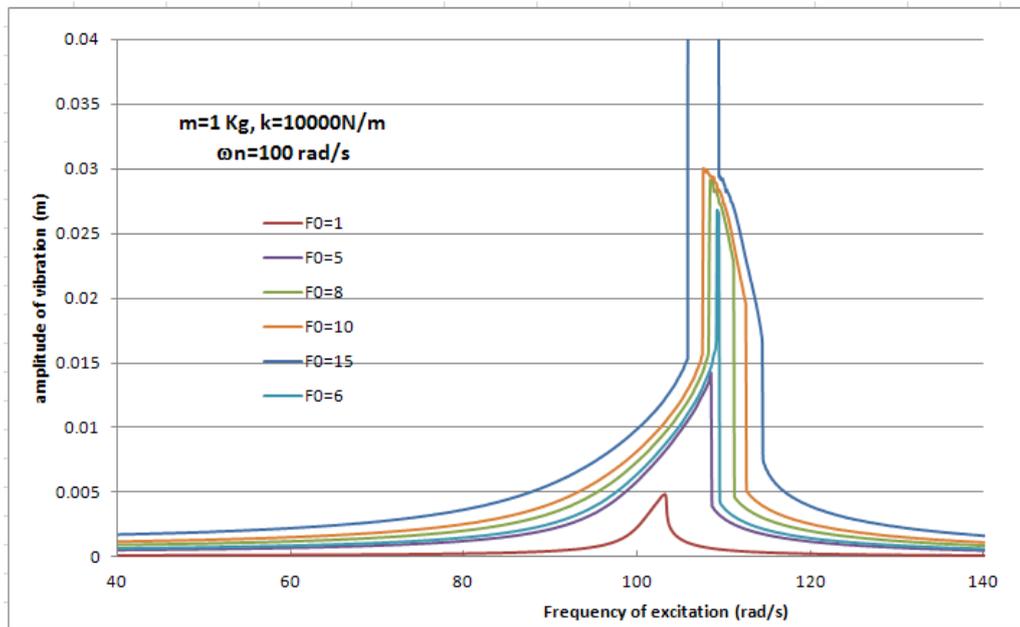


Figure 12. Amplitude versus Frequency of excitation for different excitation amplitude for different magnitude of excitation force, damping ratio maintain constant $\zeta=0.01$

Interestingly all the plots have a common point at the main system natural frequency with constant amplitude of vibration (0.006m). Moreover, as it can be seen clearly for each set of data there is an excitation frequency which corresponds to two amplitudes of vibration. At the second stage, nonlinear stiffness was added to a system with constant mass ($m=1\text{Kg}$) with different stiffness. As a result, the natural frequency of the main system was not constant. Series of runs for this case have been performed; the results have been shown in Figure 10.

As it can be seen from the graph the nonlinear stiffness has the same effect on all the series and there is no abnormal result. In other words, this nonlinear spring can be used in systems with different stiffness to get the same effect which is shifting the natural frequency to higher frequencies. According to the graph in all series of data the natural frequency has been shifted about 10% of the main system natural frequency. In the third stage, the spring with nonlinearity was added to series of system with constant stiffness and different mass. In this case the natural

frequency of the main system was not constant as well. Figure 11 shows the results of these series of runs.

As it can be seen from the graph the effect of nonlinear spring was the same on all the cases and shifted the natural frequency of the main system to higher frequencies. It has been shown that the nonlinear spring can be used with different main system parameters, mass, and stiffness. However, it should be noticed that the effect of nonlinear spring for the extreme cases could be significantly different. For instance, if the main system stiffness is too high, this nonlinear spring will not work significantly or if the stiffness of the main system is too low, this nonlinear spring could totally change the behaviour of the main system which is not desirable. As the maximum stiffness of nonlinear spring is 3000N/m from the above study, it could be concluded that for natural frequency of 100 rad/s, $k = 10000\text{N/m}$ and $m = 1\text{Kg}$ will be a proper main system.

3.5.10. Defining the Excitation Force Parameters

Another important parameter in forced vibration analysis is the amplitude of excitation F_0 . The magnitude of F_0 should be limited; otherwise, the excitation force is beyond the system capacity. Also, the introduced nonlinear spring has a limited working range ($\pm 0.03\text{m}$, see Figure 4-5). To maintain in the specific limits a series of runs has been performed for a system with $k = 10000\text{N/m}$, $m = 1\text{Kg}$ and $\zeta = 0.01$ at various F_0 . Figure 12 shows that if the F_0 goes beyond 15 N the amplitude of vibration for some frequency near the natural frequency goes to infinity. As a result, the $F_0 = 5\text{N}$ has been chosen to ensure the system does not fail in the specific frequency.

3.5.11. Results when Adding Stiffness

According to the parametric study, the main system parameters; mass $m = 1\text{Kg}$, and stiffness $k = 10000\text{N/m}$ has been chosen; The simulation results for amplitude of excitation force $F_0 = 5\text{N}$, 10N , 15N , 20N , and damping ratio $\zeta = 0.01$, 0.02 , 0.05 , 0.1 has been done, respectively and the behaviour of the main system has been studied in two different conditions. In the first scenario the main system was under an excitation force with constant amplitude, F_0 , constant light damping, ζ , and variable frequency from 40rad/s to 140 rad/s . In the second scenario, a nonlinear spring has been added in parallel to the main system as an absorber. Then, the results of a system with non-linear stiffness compared with the results of the main system (without the absorber). Figure 13 to Figure 14 show the results of simulation for the main system with and without the nonlinear spring. According to the graphs the nonlinear spring will cause to shift the frequency by about 10%. Also adding a hardening nonlinear spring will not make the system stiffer in the frequencies lower than the natural frequency of the main system. Moreover, the nonlinear spring decreases significantly the amplitude of vibration at the natural frequency of the main system in the systems with very low damping and low excitation force. In addition, in the response of the system with nonlinear spring, there is an excitation frequency that provides two different amplitudes of vibration in the same frequency (one of them is the maximum amplitude) which is an expected response of a hardening spring. In all cases the excitation force and damping ratio has been chosen to maintain the system in the allowable range.

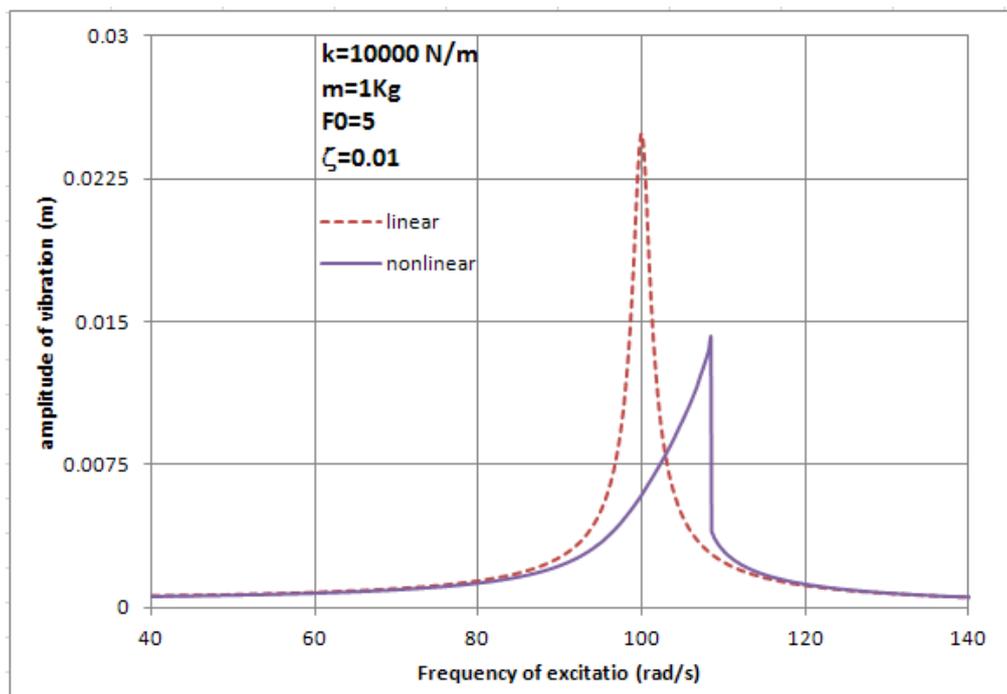


Figure 13. Amplitude verses frequency of excitation for the main system with and without nonlinear stiffness. $K=10000\text{N/m}$, $m=1\text{Kg}$, $\zeta=0.01$ and $F_0=5\text{N}$

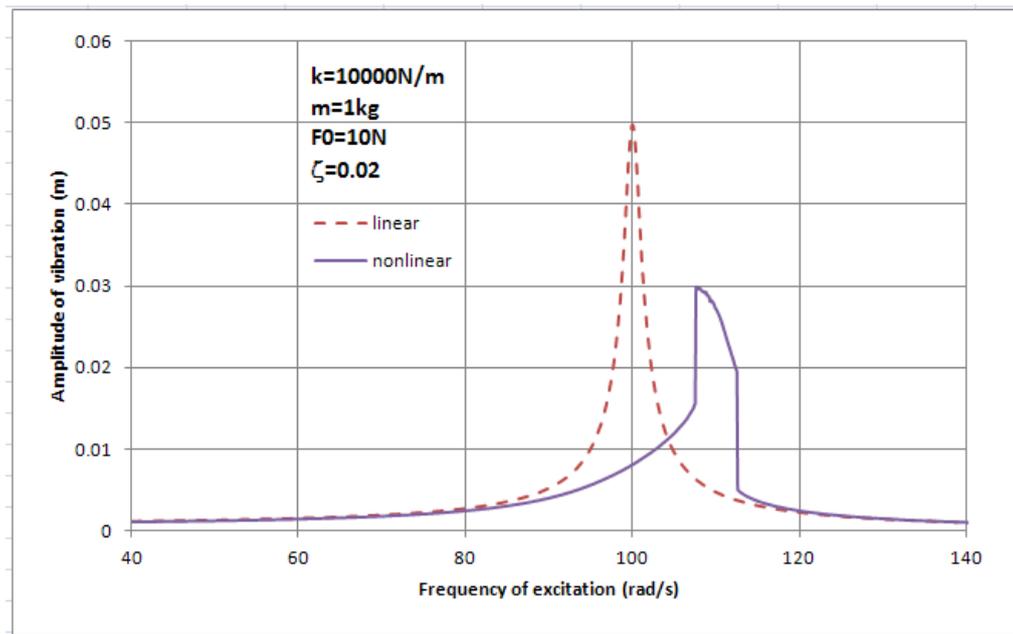


Figure 14. Amplitude versus frequency of excitation for the main system with and without nonlinear stiffness. $K=10000\text{N/m}$, $m=1\text{Kg}$, $\zeta=0.02$ and $F_0=10\text{N}$

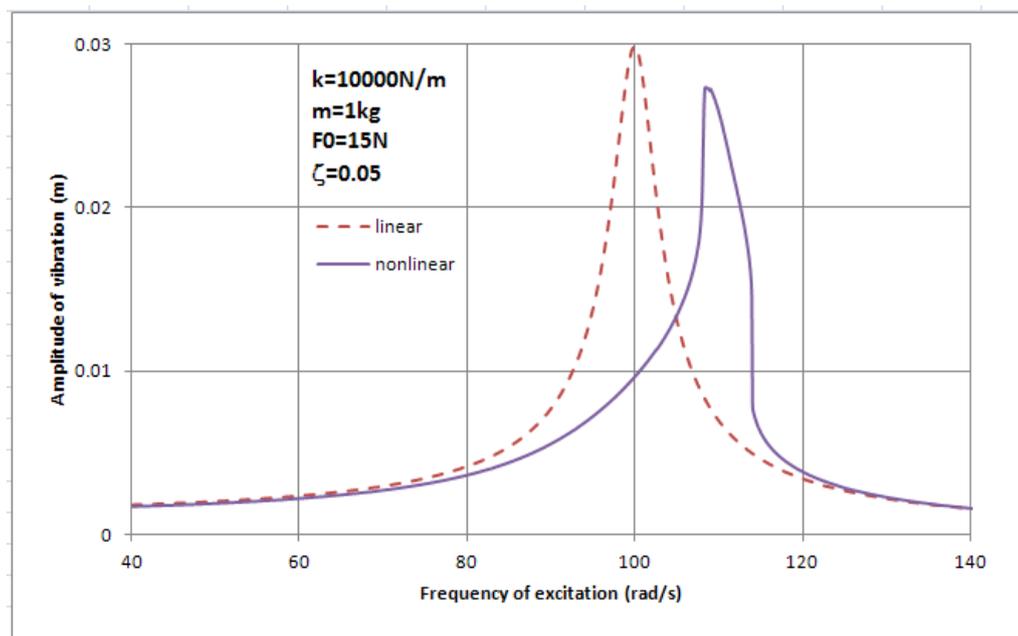


Figure 15. Amplitude versus frequency of excitation for the main system with and without nonlinear stiffness. $K=10000\text{N/m}$, $m=1\text{Kg}$, $\zeta=0.05$ and $F_0=15\text{N}$

3.5.12. Results of the Analysis for an Unbalance in Rotating Machines

Commonly, the excitation in a vibrating system is produced by an unbalance mass in rotating machinery. In this case the unbalance is represented by an unbalance mass with eccentricity of e which is rotating with angular velocity of ω . In this case, F_0 is not anymore constant and replaced with the $emu \omega^2$. In this case the response of the system with and

without the nonlinear spring has been compared as well. In the unbalance rotating machine as the amplitude of the excitation force is changing by frequency of excitation, $F_0 = mu\omega^2$, the unbalance parameters should be designed in a way to maintain the system in a working range of nonlinear spring. As in the natural frequency of the system, the excitation has maximum affect; suppose F_0 is 12N at natural frequency of the system $\omega_n = 100\text{rad/s}$ then $emu = 0.0004$ or $mu = 0.01\text{Kg}$ and $e = 0.04\text{m}$.

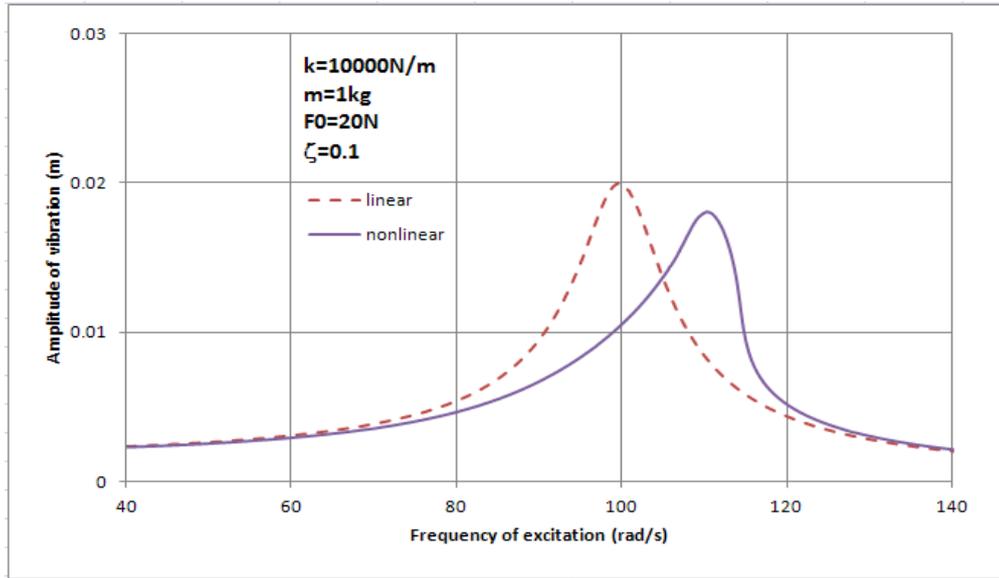


Figure 16. Amplitude versus frequency of excitation for the main system with and without nonlinear stiffness. $K=10000\text{N/m}$, $m=1\text{Kg}$, $\zeta=0.1$ and $F_0=20\text{N}$

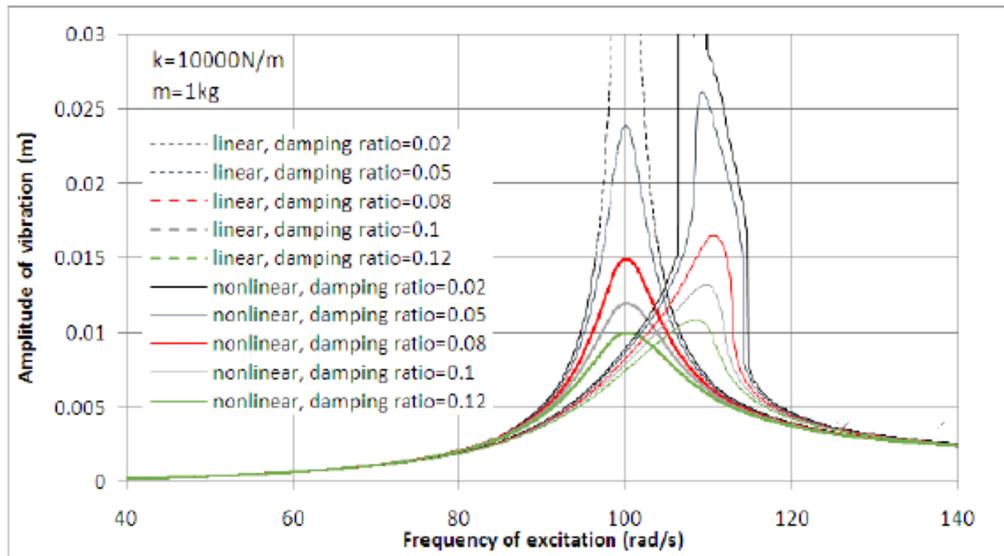


Figure 17. Amplitude of vibration versus the frequency of excitation, for the main system $K=10000\text{N/m}$, $m=1\text{Kg}$, $m_u=0.03\text{Kg}$ and $e=0.04$

3.5.13. Modelling of Non-Linear Vibration System

To verify the results of simulation a test setup has been introduced. Details have been explained in the next chapter. One of the cases has been chosen to be tested in a test rig. The system consists of unbalanced motor, main beam and magnetic spring so it is simplified to a single degree of freedom system. The unbalance mass on the motor has been selected in a way that the amplitude of excitation force will now cause out of range results for the system. For instance to exert a force with the amplitude of $F_0 = 12\text{N}$ at the spring natural frequency $\omega=100\text{rad/s}$; an unbalance mass, $m_u = 0.03\text{Kg}$ and eccentricity of $e=0.04\text{m}$ will be needed as $F_0 = e m_u \omega^2 = 0.03 * 0.04 * 100^2 = 12\text{N}$. distinct differences observed in analysing the parametric study. Firstly, it is observed that the resonance of the system is shifted away from about ten percent when a non-linear component is

added to vibration system. Secondly, the behaviour of the system response curve has changed due to the nonlinearity, that curved the system’s high amplitude and forcing it down. The resonances are detected at a higher frequency in non-linear model.

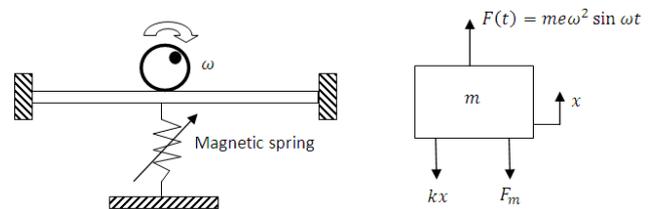


Figure 18. Proposed system schematic and free body diagram

To validate the results of the simulation an experimental setup has been built for a forced vibration system due to

unbalance mass. In the next chapter it has been shown how the experimental set up has been designed. The information obtained in the numerical analysis will be used in designing the experimental setup especially in determining the domain and resolution of sensing and data acquisition equipment.

The beam is considered as a linear spring based on fixed-fixed ends beam:

$$k_{beam} = \frac{192EI}{L^3}$$

The equivalent mass of the system = total mass +beam equivalent mass.

4. Discussion

While the study of SDOF vibration systems with non-linearity is of significant importance, there are some limitations to using this approach. Some of the limitations include:

1. Limited accuracy: SDOF models are limited in their accuracy since they only represent a simplified version of the actual system. Non-linearities can significantly affect the behavior of the system, and a single degree of freedom may not be sufficient to capture all the relevant dynamics.
2. Limited applicability: SDOF models are only applicable to systems with a single degree of freedom. However, many real-world structures have multiple degrees of freedom, which makes it difficult to accurately model their behavior using SDOF models.
3. Non-linearities are often complex: Non-linear behavior can be complex, and it is challenging to accurately model and analyze the behavior of a system with non-linearity using SDOF models.

To overcome these limitations and improve the understanding of the behavior of SDOF vibration systems with non-linearity, some suggestions for future research include:

1. Using more sophisticated models: One approach to overcome the limitations of SDOF models is to use more sophisticated models that can capture the behavior of multi-degree of freedom systems and the complexities of non-linearities.
2. Experimental validation: Conducting experiments to validate the accuracy of SDOF models can provide insight into the limitations of the approach and help identify areas for improvement.
3. Developing new analytical techniques: Developing new analytical techniques that can capture the non-linear behavior of systems with greater accuracy is another potential area for future research.
4. Integrating with other models: Integrating SDOF models with other models, such as finite element models, can provide a more comprehensive understanding of the behavior of a system with

non-linearity.

In conclusion, while SDOF models have limitations, they are still useful for gaining insights into the behavior of vibration systems with non-linearity. Future research can focus on improving the accuracy and applicability of SDOF models and integrating them with other models to provide a more comprehensive understanding of the behavior of complex systems.

5. Conclusions

In *this research paper* a single degree of freedom systems with linear undamped vibration and non-linear undamped vibration were modelled and numerically simulated. The programming environment chosen for the simulation is MATLAB, due to its capabilities to simulate the models numerically and yields the results graphically. The reliability of the model *is not* only initially tested but verified by determining the *original* frequency at where the resonance occurs in linear vibration system. The correlations gained between the simulation *and* theoretical calculation *is an experiment* that allows the code of *authenticity* to be used further to simulate the linear and non-linear undamped vibration models. A parametric study has been performed to find out how the nonlinear spring is responding to the forced vibration amplitude in the main system. There are some distinct differences observed in analysing the parametric study. Firstly, it is observed that the resonance of the system is shifted away from about ten percent when a non-linear component is added to vibration system. Secondly, the behaviour of the system response curve has changed due to the nonlinearity, that curved the system's high amplitude and forcing it down. The resonances are detected at a higher frequency in non-linear model. To validate the results of the simulation an experimental setup has been built for a forced vibration system due to unbalance mass. In the next chapter it has been shown how the experimental set up has been designed. The information obtained in the numerical analysis will be used in designing the experimental setup especially in determining the domain, resolution of sensing and data acquisition equipment and *for future research evaluation and publication*.

ACKNOWLEDGEMENTS

The authors thank Dr. J. Alrajhi for inspiring this research experiment.

REFERENCES

- [1] Alkhatib R., Golnaraghi M. F., "Active Structural Vibration Control: A Review". The Shock and Vibration Digest, vol. 35, no. 5, 2003, pp. 367-383.

- [2] Symans, M. D., and Constantinou, M. C., "Semi-active control systems for seismic protection of structures: A state-of-the-art review". 1999, *Eng. Struct.*, 21(6), 469–487.
- [3] Chu, S. Y., Soong, T. T., and Reinhorn, A. M., "Active, hybrid, and semi-active structural control: A design and implementation handbook". 2005, Wiley, Chichester, UK.
- [4] Casciati, F., Magonette, G., and Marazzi, F., "Technology of semiactive devices and applications in vibration mitigation". 2006, Wiley, Chichester, UK.
- [5] Housner, G. W., et al., "Structural control: Past, present and future". 1997, *J. Eng. Mech.*, 123(9), 897–971.
- [6] Spencer, B. F. Jr., and Nagarajaiah, S., "State of the art of structural control". 2003, *J. Struct. Eng.*, 129(7), 845–856.
- [7] Crede C.E., "Vibration and Shock Isolation", Wiley, New York, 1951.
- [8] Snowdon J.C., "Vibration isolation use and characterization". 1979. *Journal of the Acoustical Society of America* 66, 1245–1279
- [9] Rivin, E.I., "Passive Vibration Isolation". 2003. ASME Press, New York.
- [10] Frahm, H., "Device for Damping Vibrations of Bodies". 1911. US Patent US989958A.
- [11] Ormondroyd, J. and Den Hartog, J.P., "Theory of the dynamic vibration absorber". 1928. *Transactions of the American society of Mechanical Engineering*, Vol. 50, pp. 9-22.
- [12] Liu K. and Coppola G., "Optimal Design of Damped Dynamic Vibration Absorber For Damped Primary systems". 2010. *Transactions of the Canadian Society for Mechanical Engineering* Vol. 34, No. 1.
- [13] Winterflood J., Barber T.A., Blair D.G., "Mathematical analysis of an Euler spring vibration isolator". 2002. *Physics Letters A* 300, 131–139.
- [14] Goodwin A.J.H., "Vibration isolators". 1965. U.S. Patent No. 3,202,388.
- [15] Halwes D.R., "Vibration suppression system". 1980. U.S. Patent No. 4,236,607.
- [16] Flannelly W.G., "Dynamic anti-resonant vibration isolator". 1967. U.S. Patent No. 3,322,379.
- [17] Rita A.D., McGravey J.H., Jones R., "Helicopter rotor isolation evaluation utilizing the dynamic anti-resonant vibration isolator". 1978. *Journal of the American Helicopter Society* 23, 22–29.
- [18] Braun D., "Development of anti-resonance force isolators for helicopter vibration reduction". 1982. *Journal of the American Helicopter Society* 27, 37–44.
- [19] Braun D., "Vibration isolator particularly of the anti-resonance force type". 1988. U.S. Patent No. 4,781,363.
- [20] Iovovich V.A., Savovich M.K., "Isolation of floor machines by lever-type inertia vibration corrector". 2001. *Proceedings of the Institution of Civil Engineers: Structures and Buildings* 146, 391–402.
- [21] Corcoran P.E., Ticks G.E., "Hydraulic engine mount characteristics". 1984. *Proceedings of the 1984 Society of Automotive Engineers International Congress and Exhibition, Advances in Elastomeric Applications*, Detroit, MI, USA, SAE Paper No. 840407, pp. 29–34.
- [22] Flower W.C., "Understanding hydraulic mounts for improved vehicle noise, vibration and ride qualities". 1985. *Surface Vehicle Noise and Vibration Conference Proceedings*, Traverse City, MI, USA, SAE Paper No. 850975, pp. 123–132.
- [23] Yilmaz C., Kikushi N., "Analysis and design of passive low-pass filter-type vibration isolators considering stiffness and mass limitations". 2006. *Journal of Sound and Vibration* 293, 171–195.
- [24] Yilmaz C., Kikuchi N., "Analysis and design of passive band-stop filter-type vibration isolators for low-frequency applications". 2006. *Journal of Sound and Vibration* 291, 1004–1028.
- [25] Kolovsky M.Z., "Nonlinear Dynamics of Active and Passive Systems of Vibration Protection". 1999. Springer, Berlin.
- [26] Yuryev G.S., "Vibration Isolation of Precision Instruments". 1991. *Russian Academy of Sciences, Siberian Branch, Institute of Nuclear Physics Press, Novosibirsk*, Reprint, pp. 89–146.
- [27] Hundal M.S., Parnes P.S., "Response of a base excitation system with Coulomb and viscous friction". 1979. *Journal of Sound and Vibration* 64, 371–378.
- [28] Metwalli S.M., "Optimum nonlinear suspension system". 1985. *ASME Design Engineering Technical Conference*, Vol. A4, Cincinnati, OH, USA, 6pp.
- [29] Nayfeh T.A., Emaci E., Vakakis A.F., "Application of nonlinear localization to the optimization of a vibration isolation system". 1997. *American Institute of Aeronautics and Astronautics Journal* 35, 1378–1386.
- [30] Yu X., Zhu S.J., Liu S.Y., "Nonlinear normal modes for multi-degree-of-freedom nonlinear vibration isolation system". 2006. *Proceedings of the ASME Dynamic Systems and Control Division (Publication) DSC, IMECE2006*, Chicago, IL, USA, 5pp.
- [31] Popov G., Sankar S., "Modeling and analysis of nonlinear orifice type damping in vibration isolators". 1995. *Journal of Sound and Vibration* 183, 751–76.
- [32] Akihiro Shioi, Yu Otake, Ikumasa Yoshida, Shogo Muramatsu & Susumu Ohno (2023) Data-driven approximation of geotechnical dynamics to an equivalent single-degree-of-freedom vibration system based on dynamic mode decomposition, *Georisk: Assessment and Management of Risk for Engineered Systems and Geohazards*, 17: 1, 77-97.
- [33] Ruqi Sun, Waion Wong, Li Cheng, "Bi-objective optimal design of an electromagnetic shunt damper for energy harvesting and vibration control" 2023, *Mechanical Systems and Signal Processing* 182, 109571.